



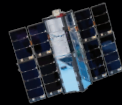
**BLUE CANYON**  
TECHNOLOGIES

SPACECRAFT BUSES, SYSTEMS & SOLUTIONS

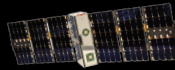
# SPACE AWAITS

With our suite of spacecraft technology and services, your team can build, test, launch and operate, all using our line of revolutionary small satellite buses and components.

## CUBESAT SOLUTIONS



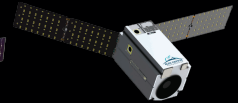
**XB3**



**XB6**



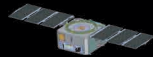
**XB12**



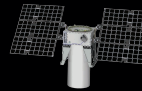
**XB16**

CLASS	3U	6U	12U	16U
AVAILABLE PAYLOAD VOLUME	1.5U (typical)	4U (typical)	8U (typical)	12U (typical)
POINTING ACCURACY	±0.003 deg (1-sigma) for 2 axes; 1 Tracker	±0.002 deg (1-sigma) 3 axes, 2 Trackers	±0.002 deg (1-sigma) 3 axes, 2 Trackers	±0.002 deg (1-sigma) 3 axes, 2 Trackers
ENERGY STORAGE	6.8 Ah	6.8 – 20.4 Ah	6.8 – 20.4 Ah	6.8 – 20.4 Ah
SOLAR ARRAY POWER	27 W	92 W - 108 W	92 W - 108 W	92 W - 108 W

## MICROSAT & MINISAT SOLUTIONS



**MERCURY CLASS**  
MICROSAT



**VENUS CLASS**  
MICROSAT



**SATURN CLASS**  
MINISAT

CLASS	11.732" Light Band	ESPA-Standard or large15" launch vehicle interface	ESPA-Grande or Equivalent 24" launch interface standard, other options available
PAYLOAD VOLUME	14.0" X 17.0" X 17.0" (launch dependent)	20.5" X 16.4" X 27.0" (1 array) 17.0" X 16.4" X 27.0" (2 array) Larger volume available depending on launch vehicle	30.0" X 30.0" X 40.0" (typical) Larger volume available within rideshare envelope and in dedicated launch vehicle fairings
POINTING ACCURACY	±0.002° (1-sigma), 3 axes, 2 Trackers		
ENERGY STORAGE	20.4 Ah	10.2 Ah	54.4 Ah
SOLAR ARRAY POWER (BOL)	SADA articulated Arrays 108 W	One wing: 222 W Two wing: 444 W	1082 W

# FEATURED COMPONENTS

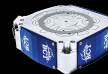
## ATTITUDE CONTROL SYSTEMS XACT-15



TYPICAL POINTING ACCURACY (1-SIGMA)  
±10 arcsec for 2 axes;  
± 25 arcsec for 3rd axis

VOLUME 10 x 10 x 5 cm (0.5U)

## REACTION WHEELS RWP500



VOLUME 110 x 110 x 38 mm

MAX TORQUE 0.025 Nm

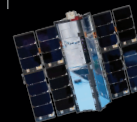
## CONTROL MOMENT GYROSCOPES CMG-12



TORQUE 12 Nm

MASS < 18 kg

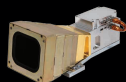
## SOLAR ARRAYS 3U



SOLAR ARRAY POWER 27 - 34 W

ARRAY VOLTAGE 14.9 VDC

## STAR TRACKERS FULL EXTENSION NST



ATTITUDE KNOWLEDGE  
Gen3: 1 asec (cross boresight);  
10 asec (around boresight)  
Gen2: 6 asec (cross boresight);  
40 asec (around boresight)

VOLUME 25 x 10 x 10 cm

# MISSIONS OPERATIONS

Our vertical integration spans from individual components to mission operations services that manage spacecraft on-orbit. Our customer-driven mission planning and on-orbit tasking allows the customer to focus on the mission while we manage the bus, leveraging our straightforward, agile interfaces.

With more than 24 years of cumulative on-orbit heritage and 50,000+ supported contacts, our Mission Operations team has the expertise you can rely on to support your mission.

# OUR MISSIONS

### ASTERIA - [NASA JPL](#)

- **Provided:** XACT ADCS System for XB6 CubeSat bus

### BLACKJACK - [Defense Advanced Research Projects Agency \(DARPA\)](#)

- **Provided:** Constellation of four Saturn Class buses

### INCUS - [Colorado State University and Jet Propulsion Laboratory](#)

- **Provided:** Constellation of three Venus Class buses

### MarCO - [NASA JPL](#)

- **Provided:** Constellation of three Venus Class buses

### METHANESAT - [MethaneSAT, LLC](#)

- **Provided:** Saturn Class bus

### RAVAN - [Johns Hopkins University Applied Physics Laboratory](#)

- **Provided:** XB3 CubeSat bus, Mission Operations

### STARLING - [NASA Ames Research Center](#)

- **Provided:** Constellation of four XB6 CubeSat buses, Mission Operations

### TEMPEST-D - [Colorado State University](#)

- **Provided:** XB6 CubeSat bus, Mission Operations

### TROPICS - [MIT Lincoln Laboratory](#)

- **Provided:** Constellation of seven XB3 CubeSat buses, Mission Operations

Note: This data is for information only and subject to change. Please contact Blue Canyon Technologies for current design data.

